Triaxial particles in shear flows : a Floquet analysis using a two-vector formulation

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- Many flows carry tiny solid objects with a very small particle Reynolds number (inertia-free limit).
- Geophysical flows : aerosols, sediments, etc.
- Industrial flows : chemical engineering, combustion, etc.
- In many cases particles are non-spherical: rods, platelets, ...complex orientational dynamics.

Axisymmetric ellipsoid or dumbbell in a simple shear flow



• In the absence of inertia and brownian motion, these particles have periodic orientations (Jeffery 1922).

• Goal:

Investigate the stability of some orbits when triaxiality is present. (See also Hinch & Leal 1979; Yarin et al. 1997).

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Motion equations

- Tiny ellipsoid with semi-axes a, b, c, in flow \vec{u} .
- Angular velocity vector with respect to the laboratory frame: $\vec{\Omega}(t)$
- Its components in a cartesian basis attached to the object:

 $\Omega_X, \Omega_Y, \Omega_Z$

• Jeffery 1922 (inertia-free object):

$$\Omega_X = \frac{b^2 g_{ZY} - c^2 g_{YZ}}{b^2 + c^2}, \qquad (1)$$

$$\Omega_Y = \frac{c^2 g_{XZ} - a^2 g_{ZX}}{c^2 + a^2}, \qquad (2)$$

$$\Omega_Z = \frac{a^2 g_{YX} - b^2 g_{XY}}{a^2 + b^2}$$
(3)

where $\mathbf{g} = \nabla \vec{u}$

Motion equations

The orientational dynamics is extremely complex: periodic motion, quasi-periodic, chaotic (Yarin et al. 1997).

$$\dot{\psi}\sin\theta\sin\phi + \dot{\theta}\cos\phi = \frac{b^2 g_{ZY} - c^2 g_{YZ}}{b^2 + c^2}, \quad (4)$$

$$\dot{\psi}\sin\theta\cos\phi - \dot{\theta}\sin\phi = \frac{c^2 g_{XZ} - a^2 g_{ZX}}{c^2 + a^2}, \quad (5)$$

$$\dot{\psi}\cos\theta + \dot{\phi} = \frac{a^2 g_{YX} - b^2 g_{XY}}{a^2 + b^2} \quad (6)$$
with $g_{ij} = g_{ij}(\theta, \psi, \phi).$

$$z$$

Complex dynamics of non-spherical objects: examples



Complex dynamics of non-spherical objects: examples



Complex dynamics of non-spherical objects: examples



Goal: re-visit the dynamics in vector form for triaxial objects.

Axisymmetric object (c = b): $\dot{\vec{e}}_X = \mathbf{B}\vec{e}_X - (\vec{e}_X \cdot \mathbf{B}\vec{e}_X)\vec{e}_X$ where **B** = $(a^2 \nabla \vec{u} - c^2 \nabla \vec{u}^T) / (a^2 + c^2)$ (Hinch & Leal 1971). $\vec{e_X}$ v x

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Axisymmetric object (c = b):

 $\vec{e}_X = \mathbf{B}\vec{e}_X - (\vec{e}_X.\mathbf{B}\vec{e}_X)\vec{e}_X$ where $\mathbf{B} = (a^2\nabla\vec{u} - c^2\nabla\vec{u}^T)/(a^2 + c^2)$ (Hinch & Leal 1971).



Starting from the rotation vector of non-inertial triaxial object, and since $\vec{e}_X = \vec{\Omega} \times \vec{e}_X$:

$$\Omega_Y = \frac{c^2 g_{XZ} - a^2 g_{ZX}}{c^2 + a^2} = -\dot{\vec{e}}_X \cdot \vec{e}_Z, \tag{7}$$

$$Q_X = \frac{a^2 g_{YX} - b^2 g_{XY}}{c^2 + a^2} = -\dot{\vec{e}}_X \cdot \vec{e}_Z, \tag{7}$$

$$\Omega_Z = \frac{a g_{YX}}{a^2 + b^2} = \vec{e}_X \cdot \vec{e}_Y$$
(8)

Leads to:

$$\dot{ec{e}}_X = rac{a^2 g_{YX} - b^2 g_{XY}}{a^2 + b^2} \, ec{e}_Y - rac{c^2 g_{XZ} - a^2 g_{ZX}}{c^2 + a^2} \, ec{e}_Z$$

Eliminate $g_{XZ} \vec{e}_Z = \nabla \vec{u}^T \vec{e}_X - g_{XX} \vec{e}_X - g_{XY} \vec{e}_Y$:

The dynamics in vector form for triaxial objects.

Leads to:

$$\left(\mathbf{S}=(
ablaec{u}+
ablaec{u}^{\,T})/2
ight)$$

$$\dot{\vec{e}}_X = \underbrace{f_1\left(\vec{e}_Y.\mathbf{S}\vec{e}_X\right)\vec{e}_Y}_{"triaxial term"} + \mathbf{B}\vec{e}_X - \left(\vec{e}_X.\mathbf{B}\vec{e}_X\right)\vec{e}_X$$

where
$$\mathbf{B} = (a^2 \nabla \vec{u} - c^2 \nabla \vec{u}^T) / (a^2 + c^2)$$

and $f_1 = \frac{2a^2(c^2 - b^2)}{(a^2 + b^2)(a^2 + c^2)}$.

Same treatment for $\dot{\vec{e}}_Y$:

$$\vec{\vec{e}}_Y = \underbrace{f_2\left(\vec{e}_Y.\mathbf{S}\vec{e}_X\right)\vec{e}_X}_{"triaxial term"} + \mathbf{C}\vec{e}_Y - \left(\vec{e}_Y.\mathbf{C}\vec{e}_Y\right)\vec{e}_Y$$

where $\mathbf{C} = (b^2 \nabla \vec{u} - c^2 \nabla \vec{u}^T) / (b^2 + c^2)$ and $f_2 = \frac{2b^2(c^2 - a^2)}{(a^2 + b^2)(b^2 + c^2)}$.

The dynamics in vector form for triaxial objects.

If

$$ec{e}_X = lpha(t) \underbrace{ec{e}_x}_{\in Lab.\ Frame} + eta(t)ec{e}_y + \gamma(t)ec{e}_z \ ec{e}_y = \delta(t) \underbrace{ec{e}_x}_{\in Lab.\ Frame} + \eta(t)ec{e}_y + \xi(t)ec{e}_z$$

then the dynamics in the plane of shear $(\alpha, \beta, \delta, \eta)$ is independent of the the *z* (vorticity) coordinates (γ, ξ) .

4 degrees of freedom (not minimal) with cubic non-linearities.

The dynamics in vector form for triaxial objects.

Set:

$$\vec{e}_X = \alpha(t)\vec{e}_x + \beta(t)\vec{e}_y + \gamma(t)\vec{e}_z, \qquad (9)$$

$$\vec{e}_Y = \delta(t)\vec{e}_x + \eta(t)\vec{e}_y + \xi(t)\vec{e}_z \qquad (10)$$

then, the exact triaxial inertia-free dynamics is:

$$\dot{\alpha} = \frac{a^2}{a^2+c^2}\beta - \frac{a^2-c^2}{a^2+c^2}\alpha^2\beta + \frac{f_1}{2}(\delta\beta+\eta\alpha)\delta, \qquad (11)$$

$$\dot{\beta} = -rac{c^2}{a^2+c^2}lpha - rac{a^2-c^2}{a^2+c^2}lphaeta^2 + rac{f_1}{2}(\deltaeta+\etalpha)\eta,$$
 (12)

$$\dot{\delta} = \frac{b^2}{b^2 + c^2} \eta - \frac{b^2 - c^2}{b^2 + c^2} \delta^2 \eta + \frac{f_2}{2} (\delta \beta + \eta \alpha) \alpha, \quad (13)$$

$$\dot{\eta} = \underbrace{-\frac{c^2}{b^2 + c^2}\delta - \frac{b^2 - c^2}{b^2 + c^2}\delta\eta^2}_{-\frac{b^2 + c^2}{b^2 + c^2}} + \frac{f_2}{2}(\delta\beta + \eta\alpha)\beta, \quad (14)$$

periodic dynamics (J. orbits)

Set:

 $ec{e}_X(t) := ec{N}_0(t) + ec{N}_1(t)$ and $ec{e}_Y(t) := ec{e}_z + ec{P}_1(t)$

With: $\vec{N}_0 = \mathbf{B}\vec{N}_0 - (\vec{N}_0 \cdot \mathbf{B}\vec{N}_0)\vec{N}_0$ (Jeffery orbit), then:

$$\vec{N}_{1} = -(\vec{N}_{0}.\mathbf{B}\vec{N}_{0})\vec{N}_{1} + f_{1}(\vec{P}_{1}.\mathbf{S}\vec{N}_{0})\vec{e}_{z} + (quad.\ terms),(15)$$

$$\dot{\vec{P}}_{1} = \mathbf{C}\vec{P}_{1} + f_{2}(\vec{P}_{1}.\mathbf{S}\vec{N}_{0})\vec{N}_{0} + (quad.\ terms)$$
(16)

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are the departure from the J. orbit.

APPLICATION: stability of orbit Y = z, for all a, b, c

Finally:

$$ec{e}_X(t) := ec{N}_0(t) + ec{N}_1(t) \qquad \textit{and} \qquad ec{e}_Y(t) := ec{e}_z + ec{P}_1(t)$$

$$\vec{N}_{1} = -(\vec{N}_{0}.\mathbf{B}\vec{N}_{0})\vec{N}_{1} + f_{1}(\vec{P}_{1}.\mathbf{S}\vec{N}_{0})\vec{e}_{z}, \qquad (17)$$

$$\dot{\vec{P}}_{1} = \mathbf{C}\vec{P}_{1} + f_{2}(\vec{P}_{1}.\mathbf{S}\vec{N}_{0})\vec{N}_{0} \qquad (18)$$

• Of the form $\dot{\vec{q}} = \mathbf{A}(t)\vec{q}$ with $\mathbf{A}(t) = \mathbf{A}(t+T)$

• Floquet theory: $\vec{q}(t) = \sum_i C_i e^{\mu_i t} \vec{p}_i(t)$, where p_i 's are *T*-periodic and $\exp(\mu_i T)$ are the eigenvalues of the monodromy matrix $\mathbf{M}(T)$ [i.e. $\dot{\mathbf{M}} = \mathbf{A}(t).\mathbf{M}(t)$, $\mathbf{M}(0) = \mathbf{I}_{\mathbf{d}}.$]

 $\rightarrow Re(\mu_i)$, if non-zero, indicates stability/instability of the orbit.



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 $a = 1, \quad b = 1, \quad c = 0.05$



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 $a = 1, \quad b = 1.2, \quad c = 0.05$





 $a = 1, \quad b = 2, \quad c = 0.05$





Conclusion: bi-vector formulation for triaxial objects.

$$\dot{\vec{e}}_{X} = \underbrace{f_{1}\left(\vec{e}_{Y}.\mathbf{S}\vec{e}_{X}\right)\vec{e}_{Y}}_{"triaxial term"} + \mathbf{B}\vec{e}_{X} - \left(\vec{e}_{X}.\mathbf{B}\vec{e}_{X}\right)\vec{e}_{X}$$

$$\dot{ec{e}}_{Y} = \underbrace{f_{2}\left(ec{e}_{Y}.\mathbf{S}ec{e}_{X}
ight)ec{e}_{X}}_{"triaxial term"} + \mathbf{C}ec{e}_{Y} - (ec{e}_{Y}.\mathbf{C}ec{e}_{Y})ec{e}_{Y}
ight] }{\mathbf{B}} = (a^{2}
abla ec{u} - c^{2}
abla ec{u}^{T})/(a^{2} + c^{2}); \ f_{1} = rac{2a^{2}(c^{2} - b^{2})}{(a^{2} + b^{2})(a^{2} + c^{2})}. \ \mathbf{C} = (b^{2}
abla ec{u} - c^{2}
abla ec{u}^{T})/(b^{2} + c^{2}); \ f_{2} = rac{2b^{2}(c^{2} - a^{2})}{(a^{2} + b^{2})(b^{2} + c^{2})}.$$

• Simple formulation, cubic non-linearities only. Allows a straightforward application of Floquet theory. Low computational cost !

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• **Perspectives:** use the bi-vect formulation to analyze chaotic dynamics (with Abdel Zalt, LUSAC-Cherbourg).

• Brownian objects, Fokker-Planck Eq. for $PDF(\vec{e}_X \bigtriangledown \vec{e}_Y) \rightarrow (a \rightarrow b \rightarrow b)$